



اللجنة الأكاديمية للهندسة المدنية

دفتر

تحليل إنشائي

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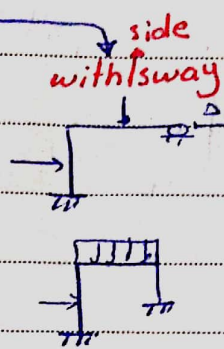
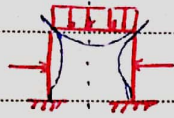
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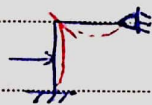
* Moment Distribution Frames.

لا يتحرك قواعدها
 without sway side without sway side with sway لا يتحرك قواعدها

1. symmetry in shape and load



لا يتحرك قواعدها
 supported against sway 2. supported against sway.

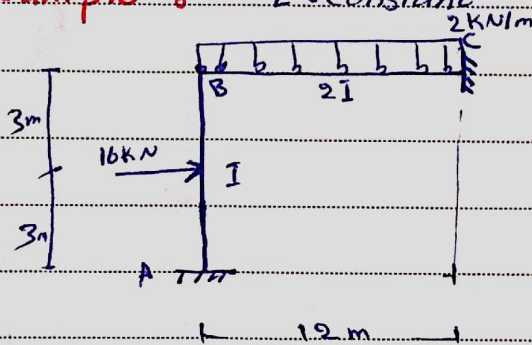


without sway

Example :- $E: \text{constant}$

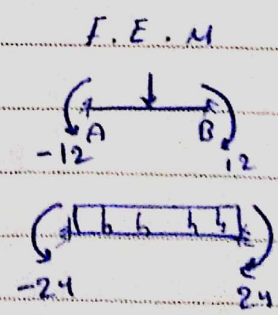
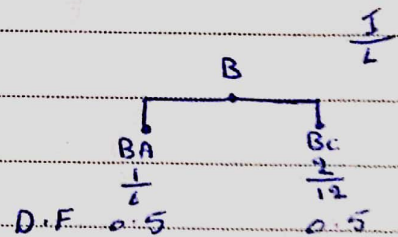
S.D and M.D ??

اول طريقة انه نحدد ال Frame و ال corner
 support نيج corner



incl truss
 th beam
 fiction
 Row

	A	I	B	C
	3m	2m	12m	
D.F	0	0.5	0.5	0
F.E.M	-12	12	-24	24
Bal	0	+6	+6	0
C.O.M	3	0	0	3
Bal	0	0	0	0
Σ Moment	-9	18	-18	24



Day: Monday

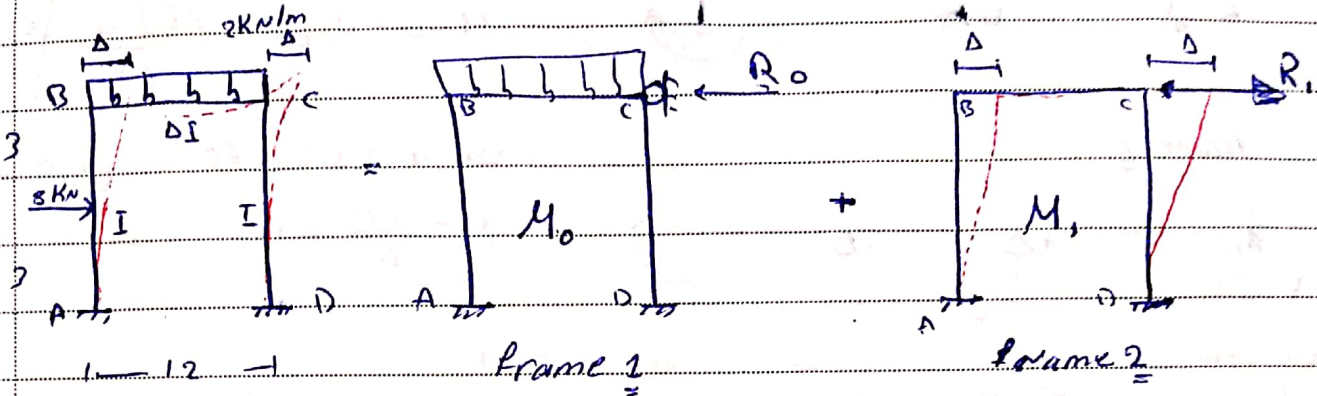
NO.

Date: 18/3/2019

Moment Distribution

Frames with sidesway

No symmetry or No support against sway



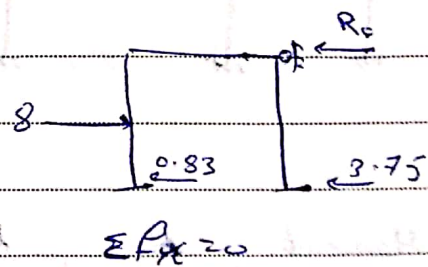
→ give displacement

$$-R_0 + \Delta R_1 = 0$$

Frame 1

Fixed corner supports

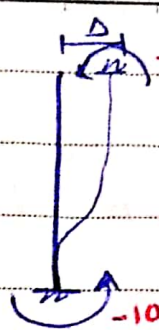
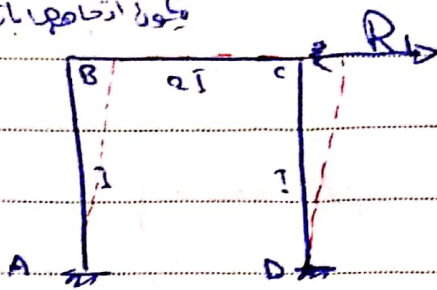
D.F.	A	B	C	D	E	F
F.E.M	-6	(6 - 24)	24	0	0	0
Ball	0	+9	+9	-12	-12	0
C.O.U	4.5	10	-6	4.5	0	-6
Ball	0	+3	+3	-2.25	-2.25	0
C.O.U	1.5	0	-1.125	1.5	0	-1.125
Ball	0	0.562	0.562	-0.75	-0.75	0
C.O.U	0.281	0	-0.281	0.281	0	-0.281
D.O.U	0	-0.1875	-0.1875	-0.1405	-0.1405	0
	6.28	1.96	-1.976	15.14	-15.14	7.4



$$R_0 + 8 - 0.83 - 3.75 - R_0 = 0$$

$$R_0 = 3.42$$

د. مكيون ارتطاف بالاس



(F.E.M)

$$M_{AB} = M_{BA} = \frac{6EI\Delta}{\delta^2}$$

$$\frac{EI\Delta}{\delta^2}$$

$$M_{CD} = M_{DC} = \frac{6EI\Delta}{\delta^2} = \frac{EI\Delta}{\delta^2}$$

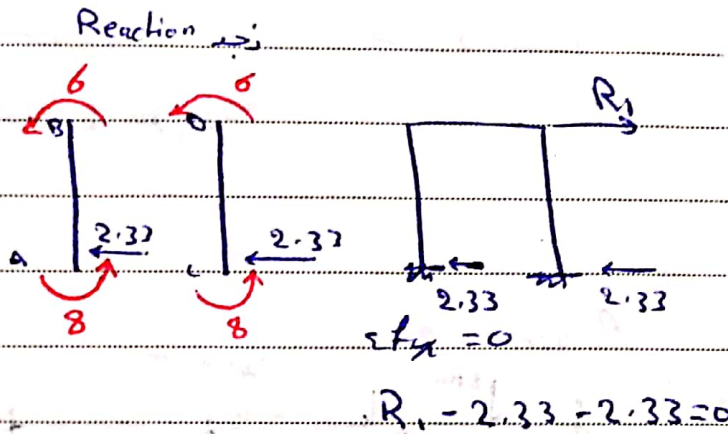
frame 2

Assume $EI\Delta = 60$

مكيون ارتطاف بالاس

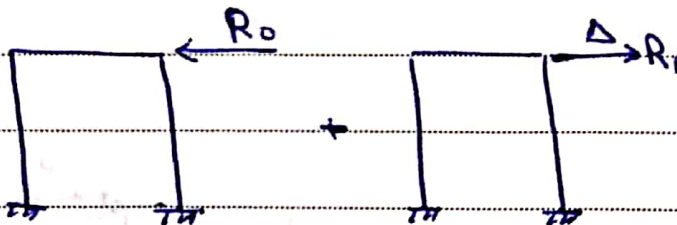
$$M = \frac{60}{6} = 10$$

	A	B	C	D	E
D.F	0	0.5	0.5	0.5	0.5
F.E.M	-10	-10	0	0	-10
Ball	0	+5	+5	+5	+5
C.O.U	2.5	0	2.5	2.5	0
Ball	0	X	X	X	X
C.O.U					
Ball					
$M_1 = \Sigma M$	-8	-6	6	+6	-8



$$R_1 - 2.33 - 2.33 = 0$$

$$R_1 = 4.66$$

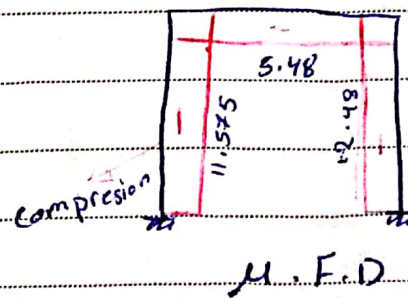
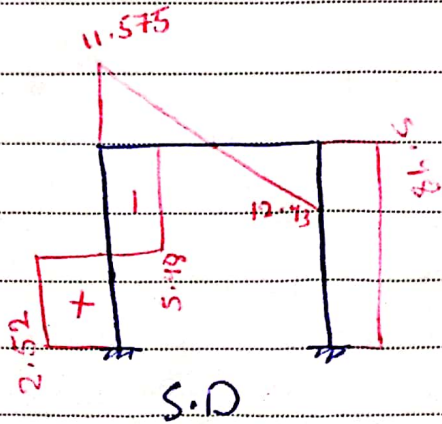
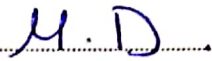


$$\Rightarrow R_0 + \Delta R_1 = 0$$

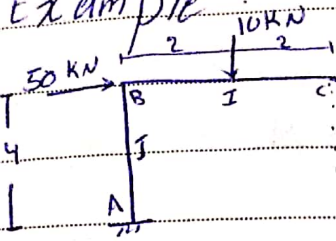
$$-3.42 + \Delta \cdot 4.66 = 0$$

$$\Delta = 0.724$$

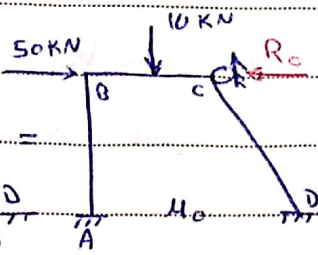
Moment	M_0	M_1	$M_{final} = M_0 + \Delta M_1$	M_{final}
M_{AB}	0.28	-8	0.28 + (0.724) \cdot -8	-5.6
M_{BA}	18.76	-6		14.4
M_{BC}	-18.76	+6		-14.4
M_{CB}	15.14	+6		19.5
M_{CD}	-15.14	-6		-19.5
M_{DC}	-7.4	-8		-13.33



Example



sway is symmetrical



$$\tan \theta = \frac{2}{4} = 0.5 \Rightarrow \theta = 26.5^\circ$$

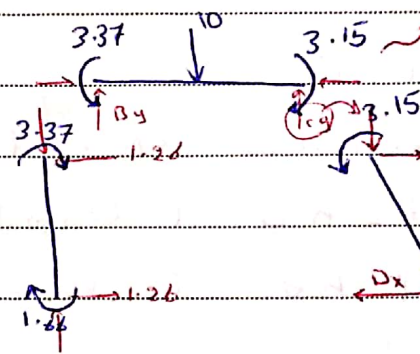
$$\Delta_{CB} = \Delta \tan \theta \quad | \quad \Delta_{CB} = 0.5 \Delta$$

$$\Delta_{CD} = \frac{\Delta}{\cos \theta} \quad | \quad \Delta_{CD} = 1.1 \Delta$$

$$\frac{\Delta_{CB}}{\Delta} = \tan \theta$$

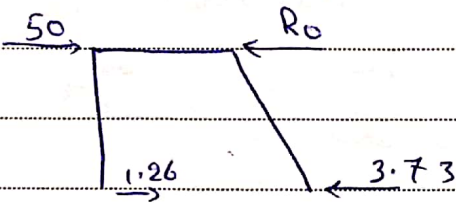
	A	B	C	D
D.F	0	0.5	0.5	0
F.E.M	0	0.5	5	0
Ball				
C.O.M				
Ball				

$$M_0 = \sum M \quad 1.66 \quad 3.37 \quad -3.37 \quad 3.5 \quad -3.15 \quad -1.55$$



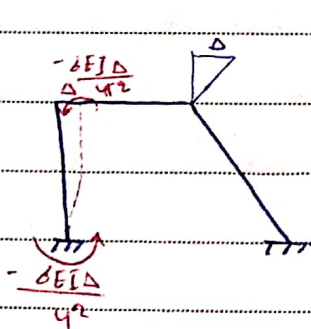
$$\sum M_C = 0$$

$$3.15 + 1.55 + 4D_x + 2 = 0 \Rightarrow D_x = -1.175$$



$$\sum F_x = 0 \quad 50 - R_0 + 1.26 - 3.73 = 0$$

$$| R_0 = 47.53$$

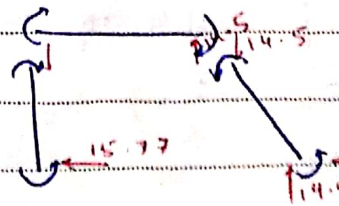


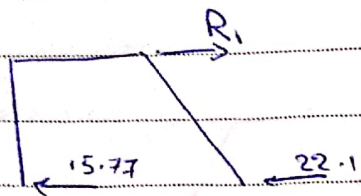
$$+ \frac{6EI(0.5\Delta)}{4^2} \quad + \frac{6EI(0.5\Delta)}{4^2}$$

$$- \frac{6EI(1.1\Delta)}{(2.05)^2} \quad - \frac{6EI(1.1\Delta)}{(2.05)^2}$$

$$\text{Assume } EID = 100$$

	A	B	C	D
D.F	0	0.5	0.5	0
F.E.M				
M_i	-37.08	-29.09	-29.09	-30.1





$$R_1 = 37.8$$

$$R_0 = \Delta R_1 \quad \dots \quad \Delta = 1.26$$

$$M = M_0 + 1.26 M_1$$

14, 15

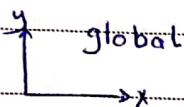
Displacement method

Stiffness Method

$$F = K \Delta \quad (\text{Normal or shear})$$

$$M = K \theta \quad (\text{Moment})$$

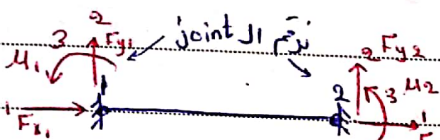
if Beam is horizontal $\Rightarrow$ global and local $\Rightarrow$ fixed



3-Degree of Freedom (3 درجات حرية)

1-D of Freedom (1 درجة حرية)

2-D of Freedom (2 درجات حرية)



each fix has 3 unknown

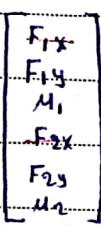
Degree of freedom

Distribution moment

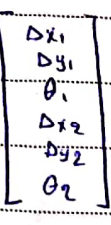
$$+ve M \Rightarrow D.C.C.W$$

Stiffness Matrix

Beam $\rightarrow$ Normal

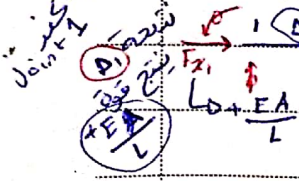


	1	2	3	4	5	6
1	$\frac{EA}{L}$	0	0	0	0	0
2	0	$\frac{12EI}{L^3}$	$\frac{6EI}{L^2}$	0	0	0
3	0	$\frac{6EI}{L^2}$	$\frac{4EI}{L}$	0	0	0
4	$-EA$	0	0	$\frac{EA}{L}$	0	0
5	0	$-\frac{6EI}{L^2}$	$-\frac{4EI}{L}$	0	$\frac{12EI}{L^3}$	$\frac{6EI}{L^2}$
6	0	$\frac{6EI}{L^2}$	$\frac{2EI}{L}$	0	$-\frac{6EI}{L^2}$	$\frac{4EI}{L}$

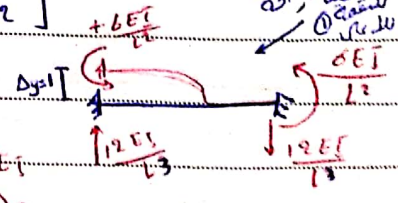
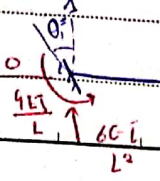


Normal $\rightarrow$ Normal force, shear force, bending moment, axial force, etc.

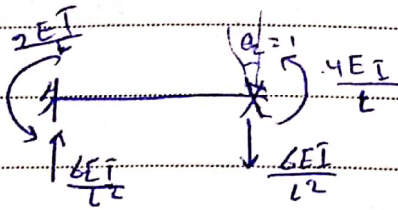
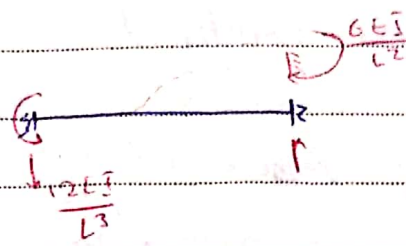
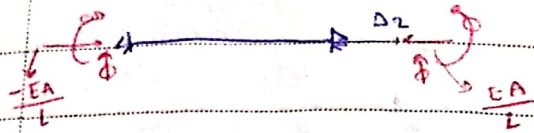
Joint 1



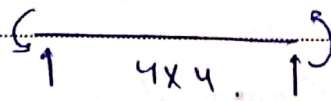
Joint 2



FIVE APPLE



⇒ Normal (log)

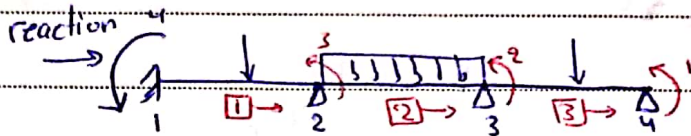


$$\begin{bmatrix} \frac{12EI}{L^3} \\ \vdots \end{bmatrix}$$

⇒ Shear (log)

2x2

$$\begin{bmatrix} \frac{4EI}{L} & \frac{2EI}{L} \\ \frac{2EI}{L} & \frac{4EI}{L} \end{bmatrix} = \frac{2EI}{L} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$



member 1, 2, 3

joint 1, 2, 3, 4

reaction 1

K member 4

$$\begin{bmatrix} 1 \end{bmatrix} \frac{2EI}{L} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$

⇒ structural stiffness

$$\begin{bmatrix} 2 \end{bmatrix} \frac{2EI}{L} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$

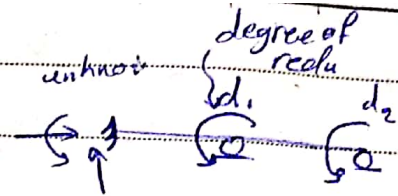
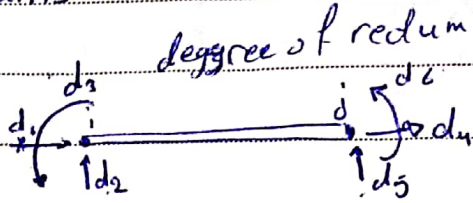
$$[F] = [K][\Delta] + [F.E.M]$$

force member

+ stiffness Method

Beams:

نصفين
الانحاء الجويت



$$K_{2 \times 2} = \frac{2EI}{L} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$

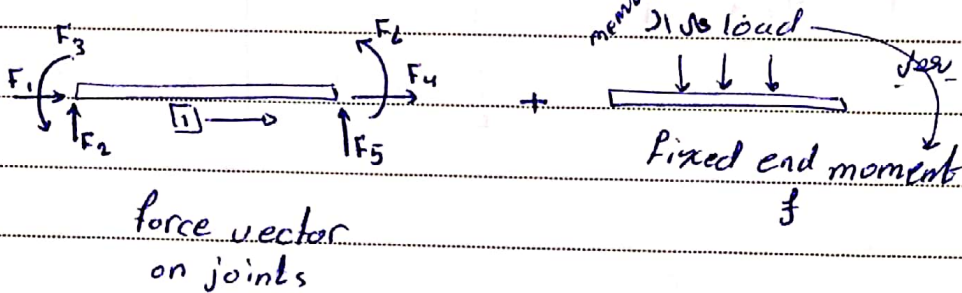
$$K_{6 \times 6}$$

البعض نادراً وجود قوى
اقتضية

$$K_{4 \times 4} \rightarrow \text{No axial}$$

self consistency

$$K_{2 \times 2} \rightarrow \text{if No axial and No shear}$$



force vector
on joints



$$[F] = [K][d] + [f]$$

structure stiffness

$$F = Kd$$

$$d = [K]^{-1} \cdot F$$

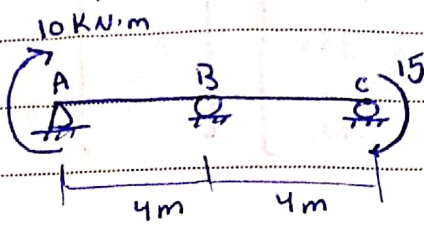
$$A = \begin{bmatrix} a & b \\ c & d \end{bmatrix} \Rightarrow A^{-1} = \frac{1}{\text{deter}} \begin{bmatrix} d & -b \\ -c & a \end{bmatrix}$$

(ad - bc)

$$\begin{bmatrix} 2 & 3 & 1 \\ 1 & 4 & 5 \\ 2 & 1 & 1 \end{bmatrix} + \begin{bmatrix} 3 \\ 5 \\ 1 \end{bmatrix} = \begin{bmatrix} \end{bmatrix}$$

المشكلة

Example



EI constant $\Rightarrow$ Draw S.D and M.D

Modeling

D.O.F. Re

الدرجة الحرة
التي يمكن
أن تحدث

force
vector

القوى على
المفاصل

joint

المفاصل

3x3

$[F]$

$\begin{bmatrix} -10 \\ 0 \\ -15 \end{bmatrix}$

member

load

المحمول

$[f]$

$\begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$

fix end moment

fix end moment

fix end moment

fix end moment

fix end moment

fix end moment

fix end moment

fix end moment

fix end moment

fix end moment

displacement matrix

$$\begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix}$$

$$[F] = [K] \cdot [d]$$

member stiffness matrix 2×2 (axial و shear في beam 1D)

$$K_1 = \frac{2EI}{4} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix}$$

$$K_2 = \frac{2EI}{4} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix}$$

structure stiffness matrix $\Sigma [K]$

$$[K]_{\text{struc}} = \begin{bmatrix} 1 & 2 & 3 \\ 1 & 0.5 & 0 \\ 0.5 & 2 & 0.5 \\ 0 & 0.5 & 1 \end{bmatrix}$$

structure stiffness equation

$$F = Kd + f^{\text{fix}}$$

$$\begin{bmatrix} -10 \\ 0 \\ -15 \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 & 0 \\ 0.5 & 2 & 0.5 \\ 0 & 0.5 & 1 \end{bmatrix} \begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix}$$

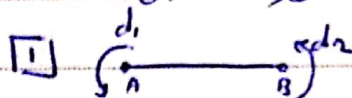
$$\begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 1 & 0.5 & 0 \\ 0.5 & 2 & 0.5 \\ 0 & 0.5 & 1 \end{bmatrix}^{-1} \begin{bmatrix} -10 \\ 0 \\ -15 \end{bmatrix}$$

$$\begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 1.1666 & -0.333 & 0.1666 \\ -0.333 & 0.166 & -0.333 \\ 0.1666 & -0.333 & 1.1666 \end{bmatrix} \begin{bmatrix} +10 \\ 0 \\ -15 \end{bmatrix}$$

$$\begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -14.16 \\ 8.33 \\ -19.17 \end{bmatrix} \begin{matrix} A \\ B \\ C \end{matrix} \text{ nod.}$$

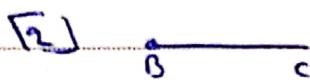
member equation

$$F = Kd + \cancel{f}$$

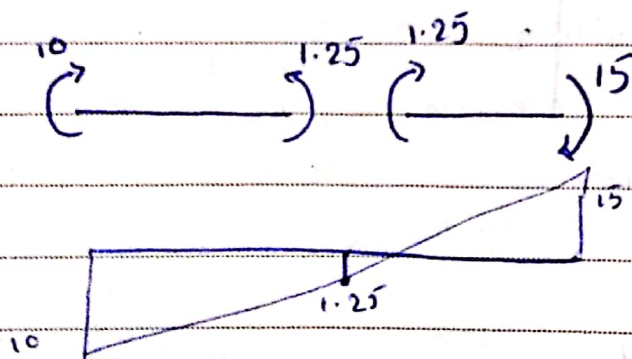


$$\begin{bmatrix} M_{AB} \\ M_{BA} \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix} * \frac{1}{EI} \begin{bmatrix} -14.16 \\ 8.33 \end{bmatrix} = \begin{bmatrix} -10 \\ 1.25 \end{bmatrix}$$

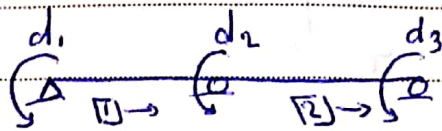
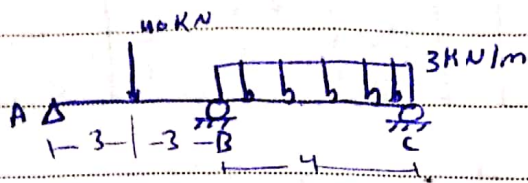
must be



$$\begin{bmatrix} M_{BC} \\ M_{CB} \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix} \begin{bmatrix} 8.33 \\ 19.17 \end{bmatrix} = \begin{bmatrix} -1.25 \\ -15 \end{bmatrix}$$



Example



$$F = Kd + f$$

3 degree of freedom $K_{\text{str}} 3 \times 3$

$$(F) \begin{matrix} \circ & \circ & \circ \\ \nearrow & \nearrow & \nearrow \end{matrix} \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$

$$(f) \begin{matrix} \downarrow 40 \\ \leftarrow 4 \end{matrix} \begin{matrix} \downarrow 30 \\ \leftarrow 4 \end{matrix} \quad (f) = \begin{bmatrix} 30 \\ -26 \\ -4 \end{bmatrix}$$

$\text{c.w} = -30 + 4 = -26$

$[K]_{\text{member}}$

$$[K_1] = \frac{2EI}{1} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} = EI \begin{bmatrix} 0.67 & 0.33 \\ 0.33 & 0.67 \end{bmatrix}$$

$$[K_2] = \frac{2EI}{4} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix}$$

$$K = \sum K = \begin{bmatrix} 0.67 & 0.33 & 0 \\ 0.33 & 1.62 & 0.5 \\ 0 & 0.5 & 1 \end{bmatrix}$$

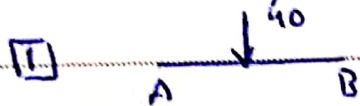
structure equation $F = Kd + f$

$$\begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} = EI \begin{bmatrix} 0.67 & 0.33 & 0 \\ 0.33 & 1.67 & 0.5 \\ 0 & 0.5 & 1 \end{bmatrix} \begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} + \begin{bmatrix} 30 \\ -26 \\ -4 \end{bmatrix}$$

$$\begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 0.67 & 0.33 & 0 \\ 0.33 & 1.67 & 0.5 \\ 0 & 0.5 & 1 \end{bmatrix}^{-1} \begin{bmatrix} -30 \\ 26 \\ 4 \end{bmatrix}$$

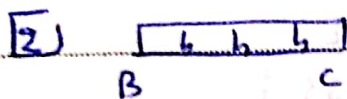
$$\begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 1.68 & -0.39 & 0.19 \\ -0.39 & 0.74 & -0.39 \\ 0.19 & -0.39 & 1.19 \end{bmatrix} \begin{bmatrix} -30 \\ 26 \\ 4 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -59.96 \\ 30.84 \\ -11.4 \end{bmatrix}$$

member equation



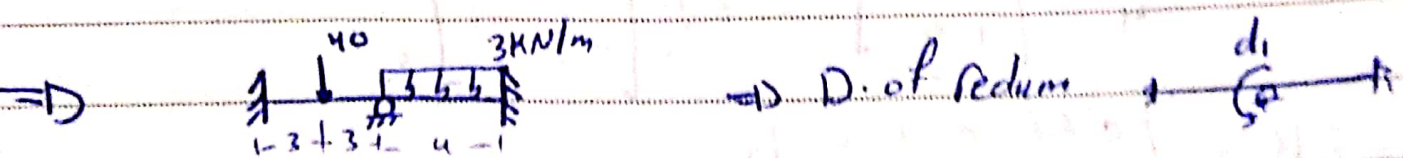
A joint = endo A is fixed and

$$\begin{bmatrix} M_{AB} \\ M_{BA} \end{bmatrix} = EI \begin{bmatrix} 0.67 & 0.33 \\ 0.33 & 0.67 \end{bmatrix} \cdot \frac{1}{EI} \begin{bmatrix} -59.96 \\ 30.84 \end{bmatrix} + \begin{bmatrix} 30 \\ -30 \end{bmatrix} = \begin{bmatrix} 0 \\ -30 \end{bmatrix}$$



$$\begin{bmatrix} M_{BC} \\ M_{CB} \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix} \cdot \frac{1}{EI} \begin{bmatrix} 30.84 \\ -11.4 \end{bmatrix} + \begin{bmatrix} 4 \\ -4 \end{bmatrix} = \begin{bmatrix} 30 \\ 0 \end{bmatrix}$$

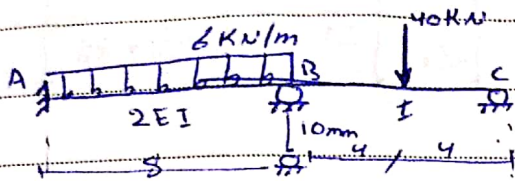
joint endo C is fixed and



Example

$$E = 200 \text{ GPa}$$

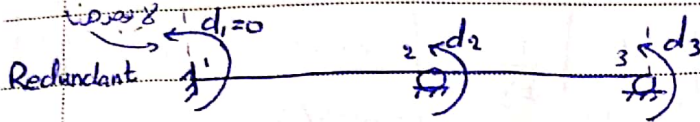
$$I = 200 \times 10^6 \text{ m}^4$$



4x4 LPLi settlement

2x2 LPLi settlement

ويفرض الاستاتيكي في



$$K = \frac{2EI}{L} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$

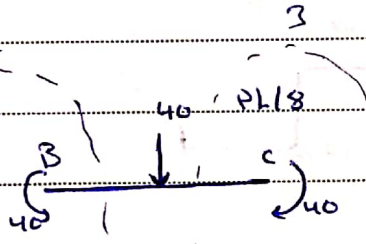
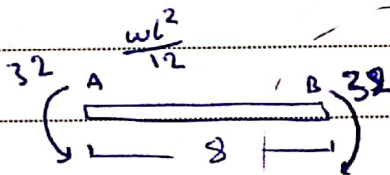
$$K_1 = EI \begin{bmatrix} 1 & 0.5 \\ 0.5 & 1 \end{bmatrix}$$

$$K_2 = EI \begin{bmatrix} 0.5 & 0.25 \\ 0.25 & 0.5 \end{bmatrix}$$

redundant

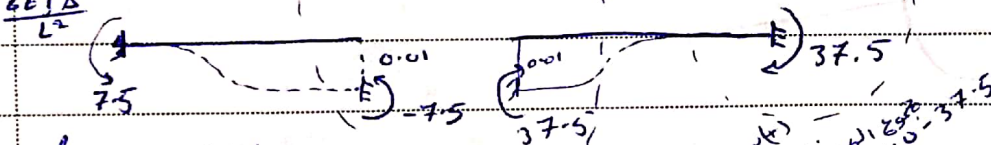
$$K = EI \begin{bmatrix} 1 & 0.5 & 0 \\ 0.5 & 1.5 & 0.25 \\ 0 & 0.25 & 0.5 \end{bmatrix}$$

Fix End moment



$$K = EI \begin{bmatrix} 1.5 & 0.25 \\ 0.25 & 0.5 \end{bmatrix}$$

$\frac{6EI\Delta}{L^2}$



forces on joint

$$F = Kd + f$$

$$\begin{bmatrix} 0 \\ 0 \end{bmatrix} = EI \begin{bmatrix} 1.5 & 0.25 \\ 0.25 & 0.5 \end{bmatrix} \begin{bmatrix} d_2 \\ d_3 \end{bmatrix} + \begin{bmatrix} 45.5 \\ -77.5 \end{bmatrix}$$

$$\begin{bmatrix} d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} -45.5 \\ 77.5 \end{bmatrix} \Rightarrow \begin{bmatrix} d_2 \\ d_3 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -61 \\ 185.45 \end{bmatrix} \text{ rad}$$

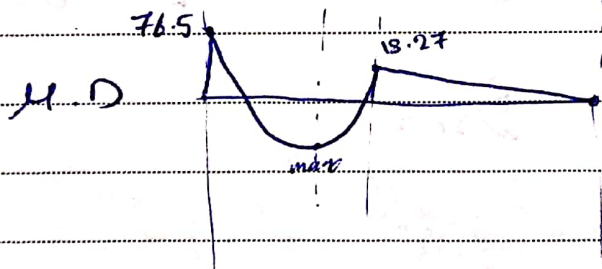
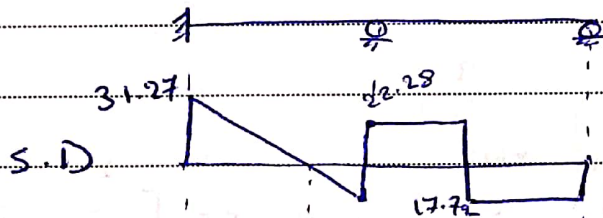
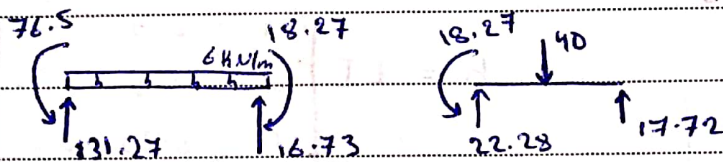
AB

$$\begin{bmatrix} M_{AB} \\ M_{BA} \end{bmatrix} = EI \begin{bmatrix} 1 & 0.5 \\ 0.6 & 1 \end{bmatrix} \begin{bmatrix} 0 \\ -61 \end{bmatrix} \frac{1}{EI} + \begin{bmatrix} 32+75 \\ -32+75 \end{bmatrix} = \begin{bmatrix} 76.5 \\ -18.27 \end{bmatrix}$$

BC

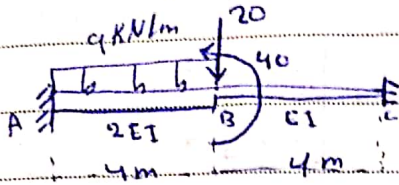
$$\begin{bmatrix} M_{BC} \\ M_{CB} \end{bmatrix} = EI \begin{bmatrix} 0.5 & 0.25 \\ 0.25 & 0.5 \end{bmatrix} \begin{bmatrix} -61 \\ 185.45 \end{bmatrix} \frac{1}{EI} + \begin{bmatrix} 40-37.5 \\ -40-37.5 \end{bmatrix} = \begin{bmatrix} 18.27 \\ 0 \end{bmatrix}$$

= 0.5 kN/m, pin at 0.25 m



Example

المسألة هي حل مسألة settlement و solution

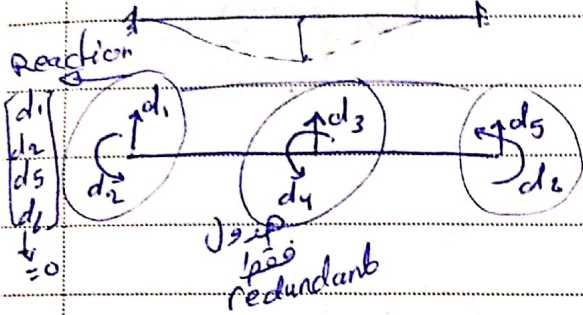


Exend moment

نحتاج إلى إيجاد الدالة و نحتاج إلى إيجاد الدالة

matrix of

moments virtual up to 4x4



$$K = EI \begin{bmatrix} 12/L^3 & 6/L^2 & -12/L^3 & 6/L^2 \\ 6/L^2 & 4/L & -6/L^2 & 2/L \\ -12/L^3 & -6/L^2 & 12/L^3 & -6/L^2 \\ 6/L^2 & 2/L & -6/L^2 & 4/L \end{bmatrix}$$

$$K_1 = EI \begin{bmatrix} 0.375 & 0.75 & -0.375 & 0.75 \\ 0.75 & 2 & -0.75 & 1 \\ -0.375 & -0.75 & 0.375 & -0.75 \\ 0.75 & 1 & -0.25 & 2 \end{bmatrix}$$

$$K_2 = EI \begin{bmatrix} 0.1875 & 0.375 & -0.1875 & 0.375 \\ 0.375 & 1 & -0.375 & 0.5 \\ -0.1875 & -0.375 & 0.1875 & 0.375 \\ 0.375 & 0.5 & -0.375 & 1 \end{bmatrix}$$

$$K = \begin{bmatrix} 0.375 & 0.75 & -0.375 & 0.75 & 0 & 0 \\ 0.75 & 2 & -0.75 & 1 & 0 & 0 \\ -0.375 & -0.75 & 0.375 & -0.75 & 0.1875 & 0.375 \\ 0.75 & 1 & -0.375 & 0.5 & 0.375 & 0.75 \\ 0 & 0 & -0.375 & -0.375 & 0.1875 & -0.375 \\ 0 & 0 & 0.375 & 0.5 & -0.375 & 1 \end{bmatrix}$$

الدالة هي الدالة

الدالة هي الدالة

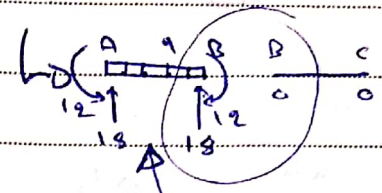
forces in joint

$$\textcircled{F} = Kd + f$$

virtual forces as redundant

fix end

$$\begin{bmatrix} -20 \\ 40 \end{bmatrix} = EI \begin{bmatrix} 0.5625 & -0.375 \\ -0.375 & 3 \end{bmatrix} \begin{bmatrix} d_3 \\ d_4 \end{bmatrix} + \begin{bmatrix} 18 \\ -12 \end{bmatrix}$$



$$\frac{1}{EI} \begin{bmatrix} K^{-1} \end{bmatrix} \begin{bmatrix} -38 \\ 52 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -61.1 \\ 9.7 \end{bmatrix}$$

deflection
Rotation

AB

$$\begin{bmatrix} A_y \\ M_{AB} \\ B_y \\ M_{BA} \end{bmatrix} = EI \begin{bmatrix} 0.375 & 0.75 & -0.375 & 0.75 \\ 0.75 & 2 & -0.75 & 1 \\ -0.375 & -0.75 & 0.375 & -0.75 \\ 0.75 & 1 & -0.75 & 2 \end{bmatrix} \begin{bmatrix} d_3 \\ d_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ -18 \\ 12 \end{bmatrix}$$

$$\begin{bmatrix} 0 \\ 0 \\ -61.1 \\ 9.7 \end{bmatrix} + \begin{bmatrix} 18 \\ 12 \\ 18 \\ -12 \end{bmatrix}$$

$$AB \Rightarrow \begin{bmatrix} 48.2 \\ 67.2 \\ -12.2 \\ 53.2 \end{bmatrix}$$

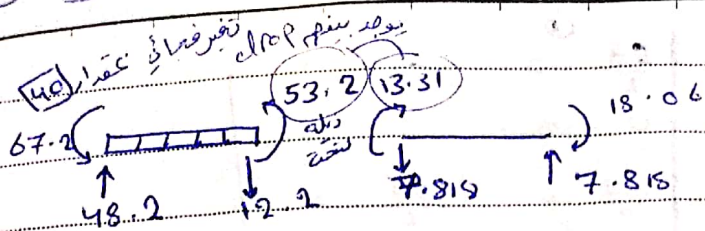
BC

$$\begin{bmatrix} B_y \\ M_{BC} \\ C_y \\ M_{CB} \end{bmatrix}$$

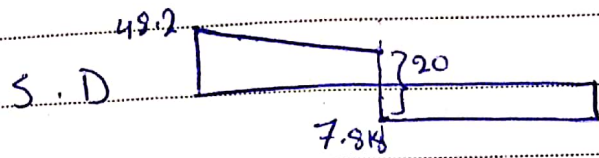
$$= EI \begin{bmatrix} 0.1875 & 0.375 & -0.1875 & 0.375 \\ 0.375 & 1 & -0.375 & 2 \\ -0.1875 & -0.375 & 0.1875 & -0.375 \\ 0.375 & 2 & -0.375 & 1 \end{bmatrix} \begin{bmatrix} d_3 \\ d_4 \end{bmatrix} + \begin{bmatrix} -61.1 \\ 9.7 \\ 0 \\ 0 \end{bmatrix}$$

$$BC \Rightarrow \begin{bmatrix} 7.818 \\ -13.31 \\ 7.818 \\ -18.06 \end{bmatrix}$$

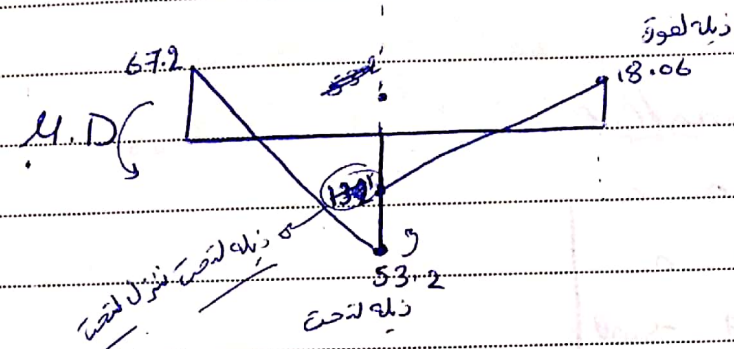
concentrated moments applied



إذا كانت M_{AB} و M_{BA}
 و M_{AB} و M_{BA}
 Reaction



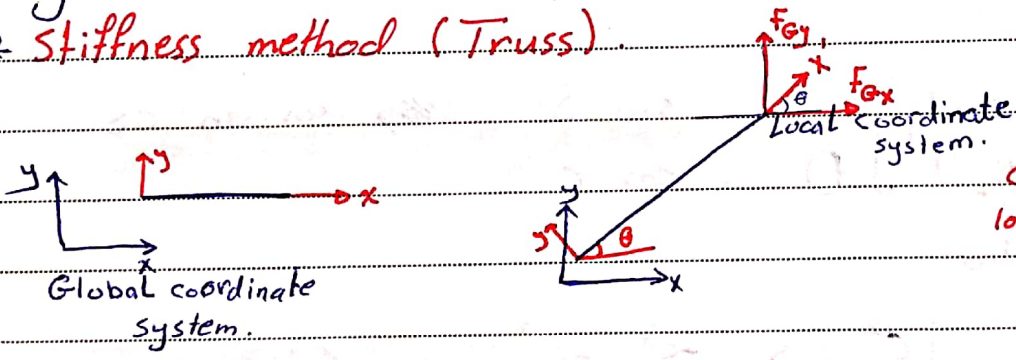
فإنه يتغير M_{AB} و M_{BA}
 M_C و M_D



Day: Monday

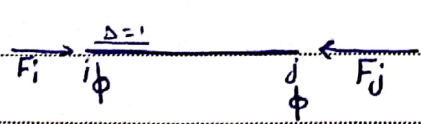
Date: 1/4/2019

#14 * Stiffness method (Truss).



Global coordinate system.
 Local coordinate system.

for horizontal member
 local = Global



$$F_i = \frac{EA}{L}$$

$$F_j = -\frac{EA}{L}$$

$$F_i = \frac{EA}{L} \phi \quad \phi \quad F_j = -\frac{EA}{L}$$

Truss stiffness matrix

$$K = \begin{array}{c|c} \textcircled{1} & \textcircled{2} \\ \hline \frac{EA}{L} & 0 \\ 0 & 0 \\ \hline -\frac{EA}{L} & 0 \\ 0 & 0 \end{array}$$

$$= \frac{EA}{L} \begin{vmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{vmatrix}$$

T: Transformation matrix.

$$T = \begin{vmatrix} \cos\theta & -\sin\theta & 0 & 0 \\ -\sin\theta & \cos\theta & 0 & 0 \\ 0 & 0 & \cos\theta & -\sin\theta \\ 0 & 0 & -\sin\theta & \cos\theta \end{vmatrix}$$

$$F_G = k_G D$$

$$F_G = \underbrace{[T K_L T^T]}_{k_G} D$$

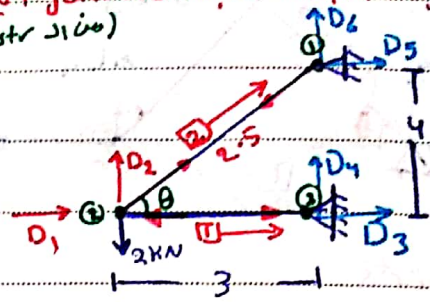
$$\begin{array}{l} \sin S \\ \cos C \end{array} \quad \begin{array}{l} \cos \cdot \sin CS \\ \cos C \end{array}$$

$$k_G = \frac{EA}{L} \begin{vmatrix} C^2 & CS & -C^2 & -CS \\ CS & S^2 & -CS & -S^2 \\ -C^2 & -CS & C^2 & CS \\ -CS & -S^2 & CS & +S^2 \end{vmatrix}$$

E 14-3 EA constant

نقطة D بالاتجاه الموجه ونبدأ من ال joint الالاستيكة (unrestr joint)

- Find Displacement at joint 2.
- find support reaction
- find forces in all member.



[1] $L=3, \theta=0, c=1, s=0, cs=0, c^2=1, s^2=0$

$K_{G1} = EA \begin{bmatrix} 0.333 & 0 & -0.333 & 0 \\ 0 & 0 & 0 & 0 \\ -0.333 & 0 & 0.333 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$

[2] $L=5$

$\theta=53.8, c=0.6, s=0.8, cs=0.48, c^2=0.36, s^2=0.64$

$K_{G2} = EA \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ 0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix}$

Support 1

Structure $K = \sum K$

$K_G = EA \begin{bmatrix} K_{11} & K_{12} & K_{13} & K_{14} & K_{15} & K_{16} \\ K_{21} & K_{22} & K_{23} & K_{24} & K_{25} & K_{26} \\ K_{31} & K_{32} & K_{33} & K_{34} & K_{35} & K_{36} \\ K_{41} & K_{42} & K_{43} & K_{44} & K_{45} & K_{46} \\ K_{51} & K_{52} & K_{53} & K_{54} & K_{55} & K_{56} \\ K_{61} & K_{62} & K_{63} & K_{64} & K_{65} & K_{66} \end{bmatrix}$

$F = \begin{bmatrix} 0 \\ -2 \\ F_3 \\ F_4 \\ F_5 \\ F_6 \end{bmatrix}$

$D = \begin{bmatrix} D_1 \\ D_2 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$

unknown displacement $F = KD$.

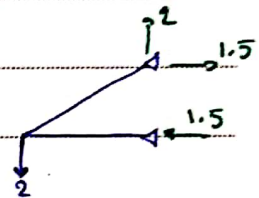
$$\begin{bmatrix} 0 \\ -2 \end{bmatrix} = EA \begin{bmatrix} 0.405 & 0.096 \\ 0.096 & 0.128 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = K^{-1} \cdot F = \frac{1}{EA} \begin{bmatrix} 4.505 \\ -19.03 \end{bmatrix}$$

support reaction.

$$K_{ur} D_u + K_{rr} D_r$$

$$\begin{bmatrix} F_3 \\ F_4 \\ F_5 \\ F_6 \end{bmatrix} = \begin{bmatrix} 3 & 0.33 & 0 \\ 4 & 0 & 0 \\ 5 & -0.072 & -0.096 \\ 6 & -0.096 & -0.128 \end{bmatrix} \begin{bmatrix} 4.505 \\ -19.03 \end{bmatrix} = \begin{bmatrix} -1.5 \\ 0 \\ 1.5 \\ 2 \end{bmatrix}$$



$$(-0.072)(4.505) - (0.096)(-19.03) = -1.5$$

$$(-0.096)(4.505) + (0.128)(19.03) = 2$$

force in all member.

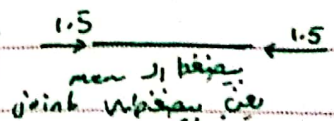
$$F_G = K D$$

member [I]

$$\begin{bmatrix} F_1 \\ F_2 \\ F_3 \\ F_4 \end{bmatrix} = K G_1 \cdot D = K G_1 \cdot \begin{bmatrix} 4.505 \\ -19.03 \\ 0 \\ 0 \end{bmatrix}$$

كل عضو من الاعضاء
يحتوي على قوتين
تسمى

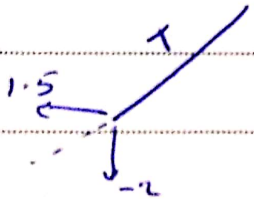
$$\begin{bmatrix} F_1 \\ F_2 \end{bmatrix} = \begin{bmatrix} 0.73 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} 4.505 \\ -19.03 \end{bmatrix} = \begin{bmatrix} 1.5 \\ 0 \end{bmatrix}$$



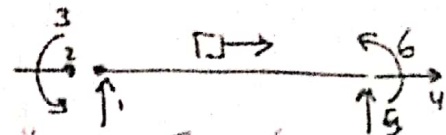
member 2

$$\begin{bmatrix} F_1 \\ F_2 \end{bmatrix} = \begin{bmatrix} 1 & 0.072 & 0.096 \\ 0 & 0.096 & 0.128 \end{bmatrix} \cdot \begin{bmatrix} 4.504 \\ -14.03 \end{bmatrix}$$

لايجاد القوة في المبر 2 $\rightarrow f = F_e \cdot T$ or $F_2 = \sqrt{1.5^2 + 2^2} = 2.5$



Frames Stiffness Matrix



$$[K^L] = \begin{bmatrix} EA/L & 0 & 0 & -EA/L & 0 & 0 \\ 0 & 12EI/L^3 & 6EI/L^2 & 0 & -12EI/L^3 & 6EI/L^2 \\ 0 & 6EI/L^2 & 4EI/L & 0 & -6EI/L^2 & 2EI/L \\ -EA/L & 0 & 0 & EA/L & 0 & 0 \\ 0 & -12EI/L^3 & -6EI/L^2 & 0 & 12EI/L^3 & -6EI/L^2 \\ 0 & 6EI/L^2 & 2EI/L & 0 & -6EI/L^2 & 4EI/L \end{bmatrix}$$

$$[T] = \begin{bmatrix} c & -s & 0 & 0 & 0 & 0 \\ s & c & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & c & -s & 0 \\ 0 & 0 & 0 & s & c & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \quad [T]^{-1} = \begin{bmatrix} c & s & 0 & 0 & 0 & 0 \\ -s & c & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & c & s & 0 \\ 0 & 0 & 0 & -s & c & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$

Member Global stiffness Matrix

Structure $K_G = \sum K$

$$[K_G] = [T][K^L][T^T] \Rightarrow K_G = \begin{bmatrix} K_{uu} & K_{ur} \\ K_{ru} & K_{rr} \end{bmatrix}$$

Structure stiffness equation:

$$[F_G]_u = [K_G]_{ur} [D_G] + [F_G]_r$$

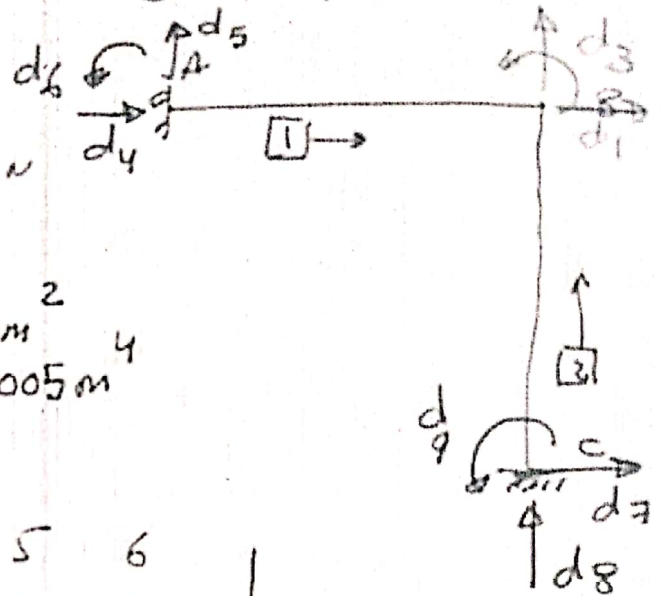
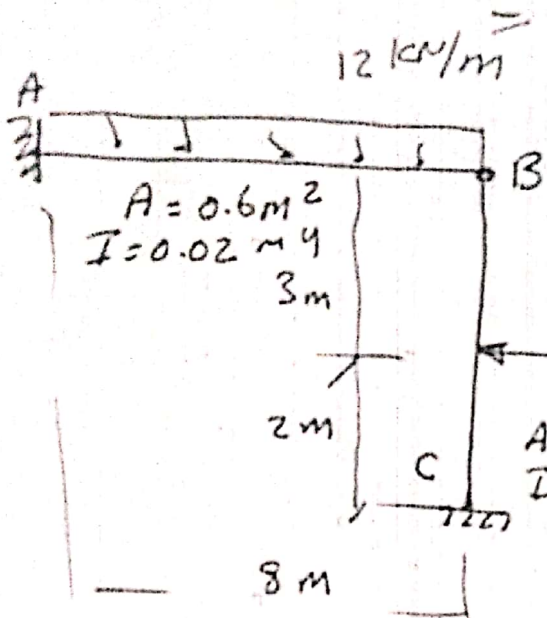
Fixed end forces

Member forces:

$$[F^L] = [K_m^L][T^T][D_G] + [T^T][F_G] \quad \text{on local coordinate system.}$$

Draw S.D, M.D.

$$E = 10^6 \text{ kN/m}^2$$



$$K_L = \begin{bmatrix} EA/L & 0 & 0 & -EA/L & 0 & 0 \\ 0 & 12EI/L^3 & 6EI/L^2 & 0 & -12EI/L^3 & 6EI/L^2 \\ 0 & 6EI/L^2 & 4EI/L & 0 & -6EI/L^2 & 2EI/L \\ -EA/L & 0 & 0 & EA/L & 0 & 0 \\ 0 & -12EI/L^3 & -6EI/L^2 & 0 & 12EI/L^3 & -6EI/L^2 \\ 0 & 6EI/L^2 & 2EI/L & 0 & -6EI/L^2 & 4EI/L \end{bmatrix}$$

Local

AB : $L=8, \theta=0, EA/L=75000, \frac{12EI}{L^3}=469, \frac{6EI}{L^2}=1875, \frac{4EI}{L}=10000, \frac{2EI}{L}=5000$
 $C=1, S=0$

$$K_1 = \begin{bmatrix} 75000 & 0 & 0 & -75000 & 0 & 0 \\ 0 & 469 & 1875 & 0 & -469 & 1875 \\ 0 & 1875 & 10000 & 0 & -1875 & 5000 \\ -75000 & 0 & 0 & 75000 & 0 & 0 \\ 0 & -469 & -1875 & 0 & 469 & -1875 \\ 0 & 1875 & 5000 & 0 & -1875 & 10000 \end{bmatrix}$$

CB : $L=5, \theta=90, EA/L=80000, \frac{12EI}{L^3}=480, \frac{6EI}{L^2}=1200, \frac{4EI}{L}=4000, \frac{2EI}{L}=2000$
 $C=0, S=1$

$$K_2 = \begin{bmatrix} 80000 & 0 & 0 & -80000 & 0 & 0 \\ 0 & 480 & 1200 & 0 & -480 & 1200 \\ 0 & 1200 & 4000 & 0 & -1200 & 2000 \\ -80000 & 0 & 0 & 80000 & 0 & 0 \\ 0 & -480 & -1200 & 0 & 480 & -1200 \\ 0 & 1200 & 2000 & 0 & -1200 & 4000 \end{bmatrix}$$

Global Stiffness Matrix

member 1

$$K_{1G} = K_{1L}$$

member 2. CB

$[K_G]$

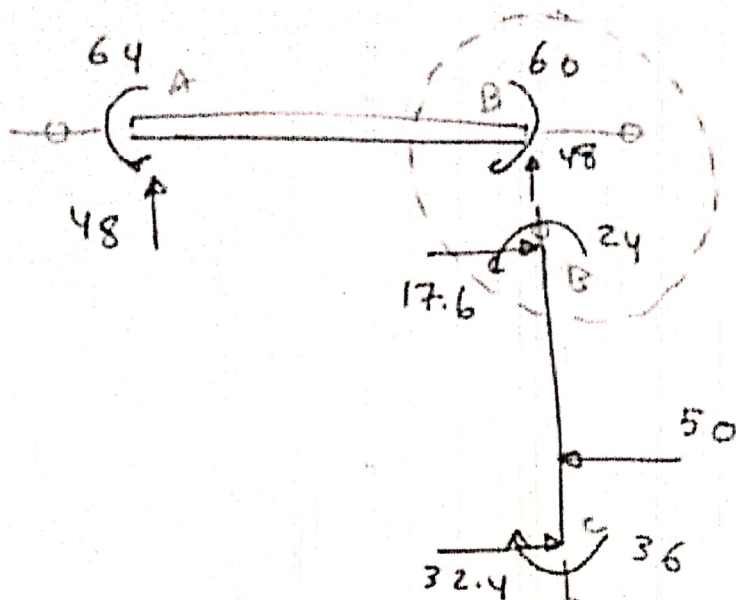
$$= \begin{bmatrix} 0 & -1 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 80000 & 0 & 0 & -80000 & 0 & 0 \\ 0 & 480 & 1200 & 0 & -480 & 1200 \\ 0 & 1200 & 4000 & 0 & -1200 & 2000 \\ -80000 & 0 & 0 & 80000 & 0 & 0 \\ 0 & -480 & -1200 & 0 & 480 & -1200 \\ 0 & 1200 & 2000 & 0 & -1200 & 4000 \end{bmatrix}$$

$$= \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ -1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 480 & 0 & -1200 & -480 & 0 & -1200 \\ 0 & 80000 & 0 & 0 & -80000 & 0 \\ -1200 & 0 & 4000 & 1200 & 0 & 2000 \\ -480 & 0 & 1200 & 480 & 0 & 1200 \\ 0 & -80000 & 0 & 0 & 80000 & 0 \\ -1200 & 0 & 2000 & 1200 & 0 & 4000 \end{bmatrix}$$

Assembly $K_G = \sum K_G$

$$K_S = \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 & 7 & 8 & 9 \\ 75480 & 0 & 1200 & -75000 & 0 & 0 & -480 & 0 & 1200 \\ 0 & 80469 & -1875 & 0 & -469 & -1875 & 0 & -80000 & 0 \\ 1200 & -1875 & 14000 & 0 & 1875 & 5000 & -1200 & 0 & 2000 \\ -75000 & 0 & 0 & 75000 & 0 & 0 & 0 & 0 & 0 \\ 0 & -649 & 1875 & 0 & 469 & 1875 & 0 & 0 & 0 \\ 0 & -1875 & 5000 & 0 & 1875 & 10000 & 0 & 0 & 0 \\ -480 & 0 & -1200 & 0 & 0 & 0 & 480 & 0 & -1200 \\ 0 & -80000 & 0 & 0 & 0 & 0 & 0 & 80000 & 0 \\ 1200 & 0 & 2000 & 0 & 0 & 0 & -1200 & 0 & 4000 \end{bmatrix}$$

Fixed End forces.



$$F_u = K_{uu} D_u + f_u$$

no forces at joint B

$$\begin{bmatrix} F_1 \\ F_2 \\ F_3 \end{bmatrix} = \begin{bmatrix} 1 & 2 & 3 \\ 75480 & 0 & 12000 \\ 0 & 80469 & -1875 \\ 12000 & -1875 & 14000 \end{bmatrix} \begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} + \begin{bmatrix} 17.6 \\ 48 \\ -40 \end{bmatrix} \Rightarrow \begin{bmatrix} d_1 \\ d_2 \\ d_3 \end{bmatrix} = \begin{bmatrix} -0.00028 \\ -0.00053 \\ 0.00281 \end{bmatrix}$$

member forces. $F_m^L = [K_m^L][T^T][D] + [T^T][f_c]$

member 1

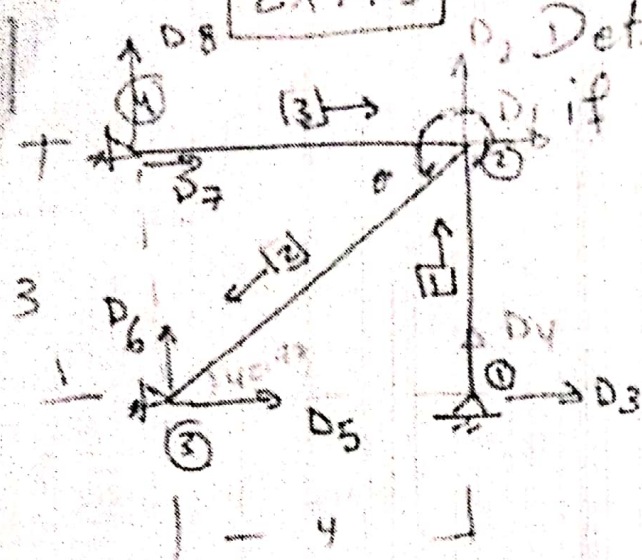
$$\begin{bmatrix} F_4 \\ F_5 \\ F_6 \\ F_1 \\ F_2 \\ F_3 \end{bmatrix} = \begin{bmatrix} K_1^L \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ -0.00028 \\ -0.00053 \\ 0.00281 \end{bmatrix} + \begin{bmatrix} 0 \\ 49 \\ 64 \\ 0 \\ 48 \\ -64 \end{bmatrix} = \begin{bmatrix} 21 \\ 53.52 \\ 73.42 \\ -21 \\ 42.5 \\ -35 \end{bmatrix}$$

Member 2. CB

$$\begin{bmatrix} F_7 \\ F_8 \\ F_9 \\ F_1 \\ F_2 \\ F_3 \end{bmatrix} = \begin{bmatrix} K_2^L \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ -0.00028 \\ -0.00053 \\ 0.00281 \end{bmatrix} + \begin{bmatrix} 0 \\ -32 \\ -36 \\ 0 \\ 17.6 \\ 2.4 \end{bmatrix} = \begin{bmatrix} 42.4 \\ -29.0 \\ -30.7 \\ -42.4 \\ -21 \\ 3.5 \end{bmatrix}$$

EX 14-5

Determine force in member 2 if support at (1) settles 25 mm
 $EA = 8 \times 10^3 \text{ kN}$



$$D = \begin{bmatrix} D_1 \\ D_2 \\ D_3 \\ D_4 \\ D_5 \\ D_6 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ -0.025 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Member 1 $L = 3, \theta = 90^\circ, C = 0, S = 1, CS = 1, C^2 = 0, S^2 = 1$

$$K_1 = EA \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0.333 & 0 & -0.333 \\ 0 & 0 & 0 & 0 \\ 0 & -0.333 & 0 & 0.333 \end{bmatrix}$$

Member 2: $L = 4, \theta = 26.87^\circ, C = -0.8, S = -0.6, CS = 0.48, C^2 = 0.64, S^2 = 0.36$

$$K_2 = EA \begin{bmatrix} 0.128 & 0.096 & -0.128 & -0.096 \\ 0.096 & 0.072 & -0.096 & -0.072 \\ -0.128 & -0.096 & 0.128 & 0.096 \\ -0.096 & -0.072 & 0.096 & 0.072 \end{bmatrix}$$

Member 3: $L = 5, \theta = 0^\circ, C = 1, S = 0, CS = 0, S^2 = 0, C^2 = 1$

$$K_3 = \begin{bmatrix} 0.25 & 0 & -0.25 & 0 \\ 0 & 0 & 0 & 0 \\ -0.25 & 0 & 0.25 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

Assembly:

$$K = EA \begin{bmatrix} 0.378 & 0.096 & 0 & 0 & -0.128 & -0.096 & -0.25 & 0 \\ 0.096 & 0.465 & 0 & -0.333 & -0.096 & -0.072 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & -0.333 & 0 & 0.333 & 0 & 0 & 0 & 0 \\ -0.128 & -0.096 & 0 & 0 & 0.128 & 0.096 & 0 & 0 \\ -0.096 & -0.072 & 0 & 0 & 0.096 & 0.072 & 0 & 0 \\ -0.25 & 0 & 0 & 0 & 0 & 0 & 0.25 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

$$F = KD$$

$$\begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \\ 7 \\ 8 \end{matrix} \begin{bmatrix} 0 \\ 0 \\ F_3 \\ F_4 \\ F_5 \\ F_6 \\ F_7 \\ F_8 \end{bmatrix} = [K] \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \\ 7 \\ 8 \end{matrix} \begin{bmatrix} D_1 \\ D_2 \\ 0 \\ -0.25 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Displacements:

$$\begin{bmatrix} 0 \\ 0 \end{bmatrix} = EA \begin{bmatrix} 0.378 & 0.096 \\ 0.096 & 0.405 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix} + \begin{matrix} 3 & 4 & 5 & 6 & 7 & 8 \\ EA \begin{bmatrix} 0 & 0 & -0.128 & -0.096 & -0.25 & 0 \\ 0 & -0.32 & -0.096 & -0.072 & 0 & 0 \end{bmatrix} \begin{bmatrix} 0 \\ -0.25 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} \end{matrix}$$

$$D_1 = 0.00556 \text{ m}$$

$$D_2 = -0.021875 \text{ m}$$

forces in Member 2.

$$\begin{bmatrix} F_1 \\ F_2 \\ F_5 \\ F_6 \end{bmatrix} = EA \begin{bmatrix} 0.128 & 0.096 & -0.128 & -0.096 \\ 0.096 & 0.072 & -0.096 & -0.072 \\ -0.128 & -0.096 & 0.128 & 0.096 \\ -0.096 & -0.072 & 0.096 & 0.072 \end{bmatrix} \begin{bmatrix} 0.00556 \\ -0.021875 \\ 0 \\ 0 \end{bmatrix}$$

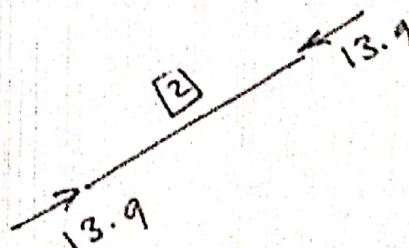
$$F_1 = -11.066$$

$$F_2 = -8.3$$

$$F_5 = 11.066$$

$$F_6 = 8.3$$

$$\Rightarrow F = \sqrt{11.066^2 + 8.3^2} = 13.9 \text{ kN}$$



Trusses with Thermal changes and Fabrication Errors

$$F_{\text{Thermal}} = (\alpha \Delta T) EA$$

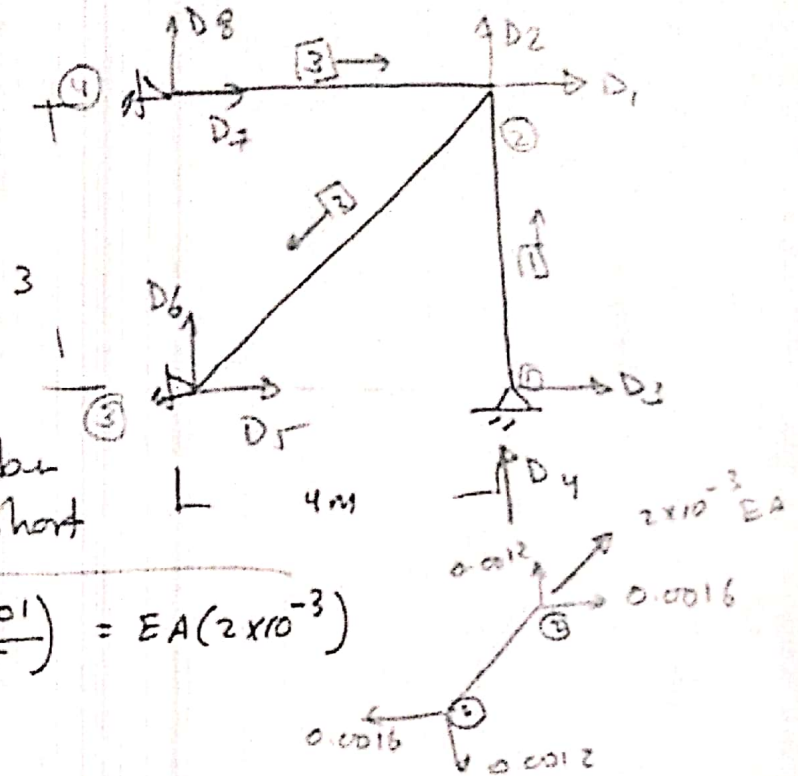
$$F_{\text{fabrication}} = \left(\frac{\Delta L}{L}\right) EA$$

$$EA = 8 \times 10^3$$

Determine the force in member 1 and 2 if member 2 was made 0.01 m too short

$$f_{\text{fabrication}} = \frac{(8 \times 10^3)}{EA} \left(\frac{-0.01}{5}\right) = EA(2 \times 10^{-3})$$

$$\begin{bmatrix} F_1 \\ F_2 \\ F_3 \\ F_4 \end{bmatrix} = EA \begin{bmatrix} 0.0016 \\ 0.0012 \\ -0.0016 \\ -0.0012 \end{bmatrix}$$



$$K_1 = EA \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0.33 & 0 & -0.33 \\ 0 & 0 & 0 & 0 \\ 0 & -0.33 & 0 & 0.33 \end{bmatrix}$$

$$K_2 = EA \begin{bmatrix} 0.128 & 0.096 & -0.128 & -0.096 \\ 0.096 & 0.072 & -0.096 & -0.072 \\ -0.128 & -0.096 & 0.128 & 0.096 \\ -0.096 & -0.072 & 0.096 & 0.072 \end{bmatrix}$$

$$K_3 = EA \begin{bmatrix} 0.25 & 0 & -0.25 & 0 \\ 0 & 0 & 0 & 0 \\ -0.25 & 0 & 0.25 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

Assembly: $K_G = \sum K$

$$K_G = EA \begin{bmatrix} 0.375 & 0.096 \\ 0.096 & 0.405 \end{bmatrix}$$

$$F_u = K_{uu} D_u + f_u$$

$$\begin{bmatrix} 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 0.375 & 0.096 \\ 0.096 & 0.405 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix} + EA \begin{bmatrix} 0.0016 \\ 0.0012 \end{bmatrix}$$

$$\Rightarrow D_1 = -0.003704 \text{ m}$$

$$D_2 = -0.002084 \text{ m}$$

Force in Member 2

$$\begin{bmatrix} P_1 \\ P_2 \end{bmatrix} = EA \begin{bmatrix} 0.128 & 0.096 \\ 0.096 & 0.072 \end{bmatrix} \begin{bmatrix} -0.003704 \\ -0.002084 \end{bmatrix} + EA \begin{bmatrix} 0.0016 \\ 0.0012 \end{bmatrix} = \begin{bmatrix} 7.41 \\ 5.55 \end{bmatrix}$$

$$f = \sqrt{7.41^2 + 5.55^2} = 9.25 \text{ kN}$$